

ON THE REDUCTION OF THE EQUATIONS OF MOTION OF A GYROHORIZONCOMPASS AND TWO-GYROSCOPE VERTICAL

(O PRIVODIMOSTI URAVNENII DVIZHENIIA GIROGORIZONTKOMPASA I DVUKHGIROSKOPICHESKOI VERTICALI)

PMM Vol. 26, No. 2, 1962, pp. 369-372

V. F. LIASHENKO
(Moscow)

(Received November 22, 1961)

Some results of [1] are generalized. The use of simplified equations of Geckeler for arbitrary motion of a gyrohorizoncompass suspension point on the earth sphere is justified with the aid of a theorem by Erugin [2]. An analogous question applicable to a two-gyroscope vertical is also considered [3].

1. In the absence of damping the equations of perturbed motion of a gyrohorizoncompass are [4]

$$mlv \frac{d\alpha}{dt} + ml \frac{dv}{dt} \alpha - mgl\beta - \Omega 2B \sin \epsilon^\circ \delta = 0 \quad \frac{d\beta}{dt} + \frac{v}{R} \alpha - \Omega \gamma = 0 \quad (1.1)$$

$$\frac{d\gamma}{dt} + \frac{2B \sin \epsilon^\circ}{mlR} \delta + \Omega \beta = 0, \quad \frac{d}{dt} (2B \sin \epsilon^\circ \delta) - mgl\gamma + mlv\Omega\alpha = 0$$

$$v = \sqrt{(RU \cos \varphi + v_E)^2 + v_N^2}, \quad \Omega = u \sin \varphi + \frac{v_E}{R} \tan \varphi - \dot{\alpha}^*, \quad \alpha^* = - \frac{v_N}{Ru \cos \varphi + v_E}$$

Here B is the kinetic moment of the gyroscope rotor; ϵ° is some equilibrium position for the separation angle of the gyroscopes; α is the deviation angle for the gyrosphere axis in azimuth; β is the lift angle for the north end of the gyrosphere axis above the surface tangent to the earth sphere at the point of gyrosphere suspension; γ is the gyrosphere rotation angle about the line north-south; δ is the gyroscopes' rotation angle about their frames, defining the perturbed location of gyro rotor axes relative to the gyrosphere; $m = P/g$ is the mass of the gyrosphere (P is the weight of the gyrosphere, g the acceleration of gravity); l is the metacentric height of the gyrosphere; R is the earth radius; U is the earth angular velocity; φ is ship's location latitude (geocentric); v_E , v_N are east and north velocity components, respectively,

of the gyrosphere suspension point relative to earth surface.

Let the ship perform maneuvers on a given latitude φ .

We will introduce new variables x_j ($j = 1, 2, 3, 4$) by formulas

$$\alpha = \frac{RU \cos \varphi}{v} x_1, \quad \beta = x_2, \quad \gamma = x_3, \quad \delta = \frac{\sin \varphi}{\sin \varphi^\circ} x_4 \quad (1.2)$$

The system (1.1) for new variables becomes [1]

$$\begin{aligned} \frac{dx_1}{dt} - \frac{v^2}{U \cos \varphi} x_2 - \lambda_1 \tan \varphi \Omega x_4 &= 0, & \frac{dx_2}{dt} + U \cos \varphi x_1 - \Omega x_3 &= 0 \\ \frac{dx_3}{dt} + \frac{2B \sin \varphi v^2}{Pl} x_4 + \Omega x_2 &= 0, & \frac{dx_4}{dt} - \frac{Pl}{2B \sin \varphi} x_3 + \frac{1}{\lambda_1} \cot \varphi \Omega x_1 &= 0 \end{aligned} \quad (1.3)$$

$$\left(v = \sqrt{\frac{g}{R}}, \lambda_1 = \frac{2Bv^2}{PlU} \right)$$

Koshliakov [1] suggested the substitution

$$\begin{aligned} \xi_1 &= x_1 \cos \theta - \frac{v}{U \cos \varphi} x_2 \cos \theta + \frac{v}{U \cos \varphi} x_3 \sin \theta - \lambda_1 \tan \varphi x_4 \sin \theta \\ \xi_2 &= \frac{U \cos \varphi}{v} x_1 \cos \theta + x_2 \cos \theta - x_3 \sin \theta - \frac{v 2B \sin \varphi}{Pl} x_4 \sin \theta \\ \xi_3 &= \frac{U \cos \varphi}{v} x_1 \sin \theta + x_2 \sin \theta + x_3 \cos \theta + \frac{v 2B \sin \varphi}{Pl} x_4 \cos \theta \\ \xi_4 &= \frac{1}{\lambda_1} \cot \varphi x_1 \sin \theta - \frac{Pl}{v 2B \sin \varphi} x_2 \sin \theta - \frac{Pl}{v 2B \sin \varphi} x_3 \cos \theta + x_4 \cos \theta \end{aligned} \quad (1.4)$$

$$(\theta(t) = \int_0^t \Omega(\tau) d\tau)$$

This substitution reduces the system (1.3) to the Schuler-Geckeler system

$$\begin{aligned} \frac{d\xi_1}{dt} - \frac{v^2}{U \cos \varphi} \xi_2 &= 0, & \frac{d\xi_3}{dt} + \frac{2B \sin \varphi v^2}{Pl} \xi_4 &= 0 \\ \frac{d\xi_2}{dt} + U \cos \varphi \xi_1 &= 0, & \frac{d\xi_4}{dt} - \frac{Pl}{2B \sin \varphi} \xi_3 &= 0. \end{aligned} \quad (1.5)$$

2. Substitution (1.4) is applicable for any form of the function $\Omega(t)$. Nevertheless, [1] presents its substantiation only for a special case when $\Omega(t)$ is a periodic function of time. Equations (1.5), however, can be justified for any form of the function $\Omega(t)$ by means of a theorem due to Erugin [2].

The system (1.3) is solved in a similar way to that suggested in [4]. Let us represent the system (1.3) in the form

$$\frac{d}{dt} \left(\frac{U \cos \varphi}{v} x_1 \right) - v x_2 - \frac{2B \sin \varphi v}{Pl} \Omega x_4 = 0, \quad \frac{dx_2}{dt} + U \cos \varphi x_1 - \Omega x_3 = 0 \quad (2.1)$$

$$\frac{dx_3}{dt} + \frac{2B \sin \varphi v^2}{Pl} x_4 + \Omega x_2 = 0, \quad \frac{d}{dt} \left(\frac{2B \sin \varphi v}{Pl} x_4 \right) - vx_3 + \frac{U \cos \varphi}{v} \Omega x_1 = 0$$

Introducing two complex-valued functions of t by

$$\chi(t) = \frac{U \cos \varphi}{v} x_1 + ix_2, \quad \mu(t) = x_3 - i \frac{2B \sin \varphi v}{Pl} x_4 \quad (i = \sqrt{-1}) \quad (2.2)$$

The system (2.1) is reduced to two equations of the form

$$\frac{d\chi}{dt} + iv\chi = i\Omega\mu, \quad \frac{d\mu}{dt} + iv\mu = i\Omega\chi \quad (2.3)$$

which yield the following equations

$$\frac{d}{dt}(\chi + \mu) + i(v - \Omega)(\chi + \mu) = 0, \quad \frac{d}{dt}(\chi - \mu) + i(v + \Omega)(\chi - \mu) = 0 \quad (2.4)$$

These are easily integrated. We have

$$\chi + \mu = C_1 \exp\left(-i \int_0^t (v - \Omega) dt\right), \quad \chi - \mu = C_2 \exp\left(-i \int_0^t (v + \Omega) dt\right) \quad (2.5)$$

Here C_1, C_2 are arbitrary constants. The general solution of system (2.3) is of the form

$$\chi = \frac{1}{2} e^{-ivt} (C_1 e^{i\theta} + C_2 e^{-i\theta}), \quad \mu = \frac{1}{2} e^{-ivt} (C_1 e^{i\theta} - C_2 e^{-i\theta}) \quad (2.6)$$

3. It follows from the solution (2.6) that the integral matrix of the system (2.3) has the structure

$$P = e^{-ivt} Z, \quad Z(t) = \frac{1}{2} \begin{vmatrix} e^{i\theta} & e^{-i\theta} \\ e^{i\theta} & -e^{-i\theta} \end{vmatrix} \quad \text{--- Liapunov type matrix} \quad (3.1)$$

It follows, therefore, that on the basis of a theorem due to Erugin [2], the system (2.3) is reducible for any form of the function $\Omega(t)$. The substitution

$$Y = Z^{-1}X, \quad Y(t) = \begin{vmatrix} y_1 \\ y_2 \end{vmatrix}, \quad Z^{-1}(t) = \begin{vmatrix} e^{-i\theta} & e^{-i\theta} \\ e^{i\theta} & -e^{i\theta} \end{vmatrix}, \quad X(t) = \begin{vmatrix} \chi \\ \mu \end{vmatrix} \quad (3.2)$$

transforms the system (2.3) into the system with constant coefficients

$$\frac{dy_1}{dt} + ivy_1 = 0, \quad \frac{dy_2}{dt} + ivy_2 = 0 \quad (3.3)$$

Inverse transformation from the variables y_1, y_2 to the variables χ, μ according to (3.2) is of the form

$$X = ZY \quad (3.4)$$

Letting

$$y_1 = \eta_1 + i\eta_2, \quad y_3 = \eta_3 + i\eta_4 \quad (3.5)$$

we have on the basis of (3.3) the Schuler-Geckeler system for the variables η_j

$$\frac{d\eta_1}{dt} - v\eta_2 = 0, \quad \frac{d\eta_2}{dt} + v\eta_1 = 0, \quad \frac{d\eta_3}{dt} - v\eta_4 = 0, \quad \frac{d\eta_4}{dt} + v\eta_3 = 0 \quad (3.6)$$

From (3.2) and considering (2.2) and (3.5), we obtain the formulas for a non-singular transformation from the variables x_j to the variables η_j of the form

$$\begin{aligned} \eta_1 &= \frac{U \cos \varphi}{v} x_1 \cos \theta + x_2 \sin \theta + x_3 \cos \theta - \frac{2B \sin \varphi v}{Pl} x_4 \sin \theta \\ \eta_2 &= -\frac{U \cos \varphi}{v} x_1 \sin \theta + x_2 \cos \theta - x_3 \sin \theta - \frac{2B \sin \varphi v}{Pl} x_4 \cos \theta \\ \eta_3 &= \frac{U \cos \varphi}{v} x_1 \cos \theta - x_2 \sin \theta - x_3 \cos \theta - \frac{2B \sin \varphi v}{Pl} x_4 \sin \theta \\ \eta_4 &= \frac{U \cos \varphi}{v} x_1 \sin \theta + x_2 \cos \theta - x_3 \sin \theta + \frac{2B \sin \varphi v}{Pl} x_4 \cos \theta \end{aligned} \quad (3.7)$$

The formulas for inverse transformation from the variables η_j to the variables x_j are

$$\begin{aligned} x_1 &= \frac{1}{2} \frac{v}{U \cos \varphi} (\eta_1 \cos \theta - \eta_2 \sin \theta + \eta_3 \cos \theta + \eta_4 \sin \theta) \\ x_2 &= \frac{1}{2} (\eta_1 \sin \theta + \eta_2 \cos \theta - \eta_3 \sin \theta + \eta_4 \cos \theta) \\ x_3 &= \frac{1}{2} (\eta_1 \cos \theta - \eta_2 \sin \theta - \eta_3 \cos \theta - \eta_4 \sin \theta) \\ x_4 &= \frac{1}{2} \frac{Pl}{v 2B \sin \varphi} (-\eta_1 \sin \theta - \eta_2 \cos \theta - \eta_3 \sin \theta + \eta_4 \cos \theta) \end{aligned} \quad (3.8)$$

4. The preceding theory is applicable virtually without any alteration to the equations of a two-gyroscope vertical as well, given in [3]:

$$\begin{aligned} mav \frac{d\alpha}{dt} + ma \frac{dv}{dt} \alpha - mga\beta + \Omega 2B \cos \theta^* \delta = 0 \quad \frac{d\beta}{dt} + \frac{v}{R} \alpha - \Omega \gamma = 0 \\ \frac{d\gamma}{dt} - \frac{2B \cos \theta^*}{maR} \delta + \Omega \beta = 0, \quad \frac{d}{dt} (2B \cos \theta^* \delta) + mga\gamma - mav\Omega \alpha = 0 \end{aligned} \quad (4.1)$$

Function $\theta^*(t)$ satisfies the condition

$$\theta^*(t) = \sin^{-1} \frac{mav}{2B}$$

The remaining notation in system (4.1) is the same as in (1.1), with α having the same meaning as l . Let us introduce new variables z_j by formulas

$$\alpha = \frac{RU \cos \varphi}{v} z_1, \quad \beta = z_2, \quad \gamma = z_3, \quad \delta = \frac{\cos \varphi}{\cos \theta^*} z_4 \quad (4.2)$$

The system (4.1) will become

$$\begin{aligned} \frac{dz_1}{dt} - \frac{v^2}{U \cos \varphi} z_2 + \lambda_2 \Omega z_4 = 0, \quad \frac{dz_3}{dt} - \frac{2B \cos \varphi v^2}{Pa} z_4 + \Omega z_2 = 0 \\ \frac{dz_2}{dt} + U \cos \varphi z_1 - \Omega z_3 = 0, \quad \frac{dz_4}{dt} + \frac{Pa}{2B \cos \varphi} z_3 - \frac{1}{\lambda_2} \Omega z_1 = 0 \end{aligned} \quad (4.3)$$

$(\lambda_2 = \frac{2Bv^2}{PaU})$

With analogous reasoning as above, one can show that by means of the non-singular substitution

$$\begin{aligned} \zeta_1 &= \frac{U \cos \varphi}{v} z_1 \cos \theta + z_2 \sin \theta + z_3 \cos \theta + \frac{2B \cos \varphi v}{Pa} z_4 \sin \theta \\ \zeta_2 &= -\frac{U \cos \varphi}{v} z_1 \sin \theta + z_2 \cos \theta - z_3 \sin \theta + \frac{2B \cos \varphi v}{Pa} z_4 \cos \theta \\ \zeta_3 &= \frac{U \cos \varphi}{v} z_1 \cos \theta - z_2 \sin \theta - z_3 \cos \theta + \frac{2B \cos \varphi v}{Pa} z_4 \sin \theta \\ \zeta_4 &= \frac{U \cos \varphi}{v} z_1 \sin \theta + z_2 \cos \theta - z_3 \sin \theta - \frac{2B \cos \varphi v}{Pa} z_4 \cos \theta \end{aligned} \quad (4.4)$$

the system (4.3) is reducible for any form of the function $\Omega(t)$ to the Schuler-Geckeler system

$$\frac{d\zeta_1}{dt} - v\zeta_3 = 0, \quad \frac{d\zeta_2}{dt} + v\zeta_1 = 0, \quad \frac{d\zeta_3}{dt} - v\zeta_4 = 0, \quad \frac{d\zeta_4}{dt} + v\zeta_2 = 0 \quad (4.5)$$

The formulas for inverse transformation from the variables ζ_j to the variables z_j are

$$\begin{aligned} z_1 &= \frac{1}{2} \frac{v}{U \cos \varphi} (\zeta_1 \cos \theta - \zeta_2 \sin \theta + \zeta_3 \cos \theta + \zeta_4 \sin \theta) \\ z_2 &= \frac{1}{2} (\zeta_1 \sin \theta + \zeta_2 \cos \theta - \zeta_3 \sin \theta + \zeta_4 \cos \theta) \\ z_3 &= \frac{1}{2} (\zeta_1 \cos \theta - \zeta_2 \sin \theta - \zeta_3 \cos \theta - \zeta_4 \sin \theta) \\ z_4 &= \frac{1}{2} \frac{Pa}{v 2B \cos \varphi} (\zeta_1 \sin \theta + \zeta_2 \cos \theta + \zeta_3 \sin \theta - \zeta_4 \cos \theta) \end{aligned} \quad (4.6)$$

BIBLIOGRAPHY

1. Koshliakov, V.N., O privodimosti uravnenii dvizhenia girogorizont-kompassa (On the reduction of the equations of motion of a gyrohorizoncompass). *PMM* Vol. 25, No. 5, 1961.
2. Erugin, N.P., *Privodimye sistemy (Reducible Systems)*. Izd-vo Akad. Nauk SSSR, 1946.

3. Ishlinskii, A.Iu., Teoria dvukhgiroskopicheskoi vertikali (Theory of a two-gyroscope vertical). *PMM* Vol. 21, No. 2, 1957.
4. Ishlinskii, A.Iu., K teorii girogorizontkompasa (On the theory of the gyrohorizoncompass). *PMM* Vol. 20, No. 4, 1956.

Translated by V.C.